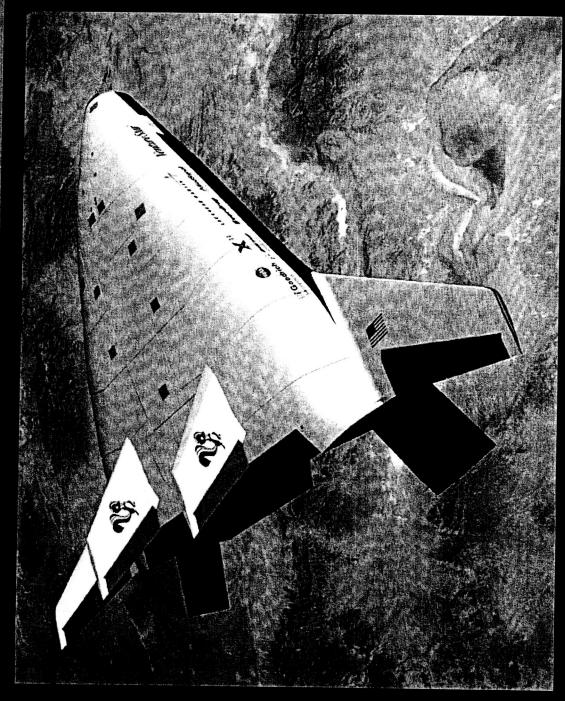
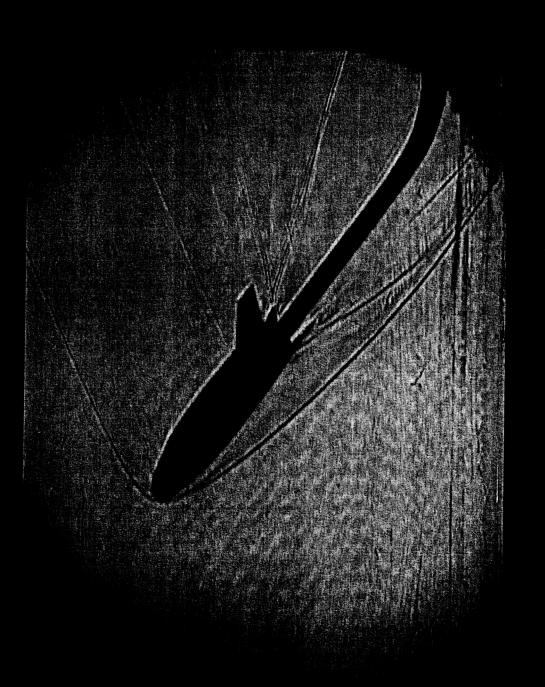


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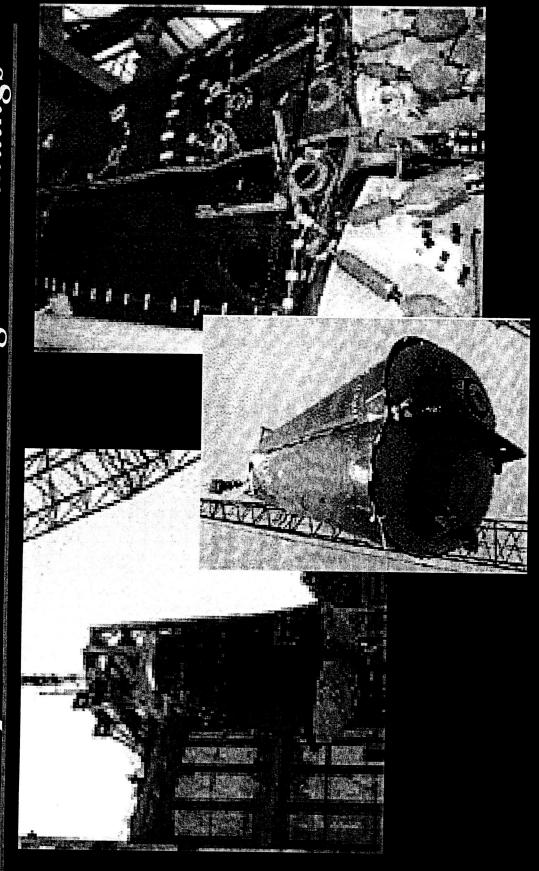
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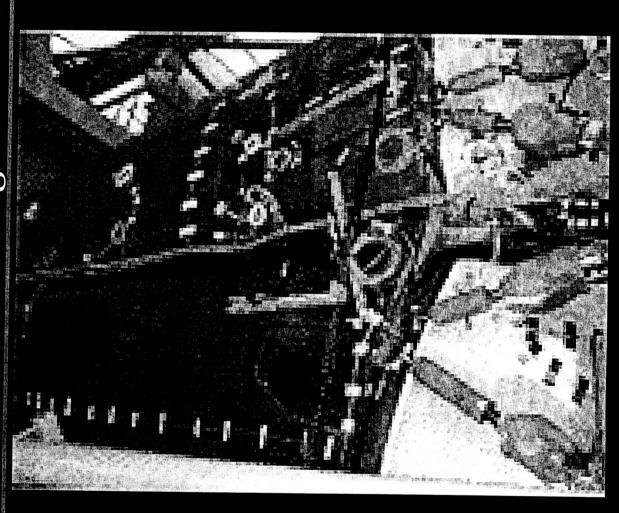




Introduction

- Tank History and Test Objectives
 - Failure Description
- Investigation
- Conclusions



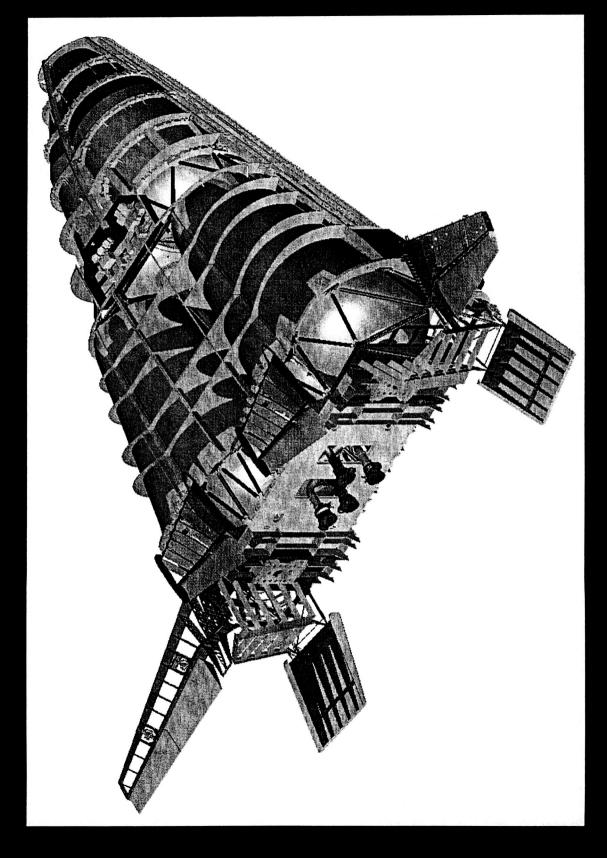


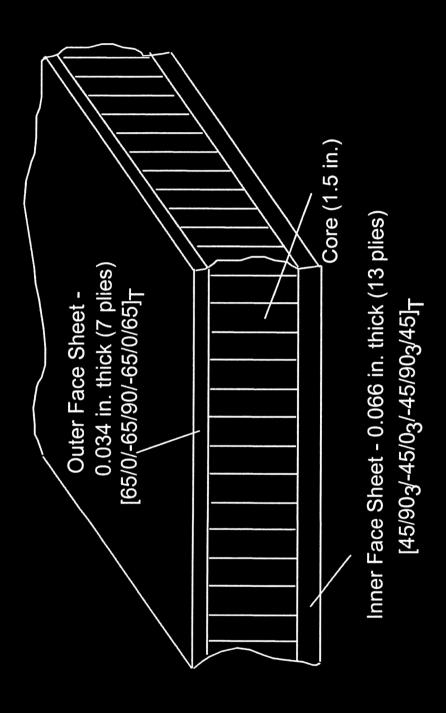
Tank Description

Structural component of the aft body

Quad-lobe design

 Sandwich - honeycomb graphite epoxy construction





Geometry of sandwich structure

Test Objectives

 Verify structural integrity at 105% expected flight load limit varying the following parameters

Cryogenic temperature

Internal pressure

Mechanical loading

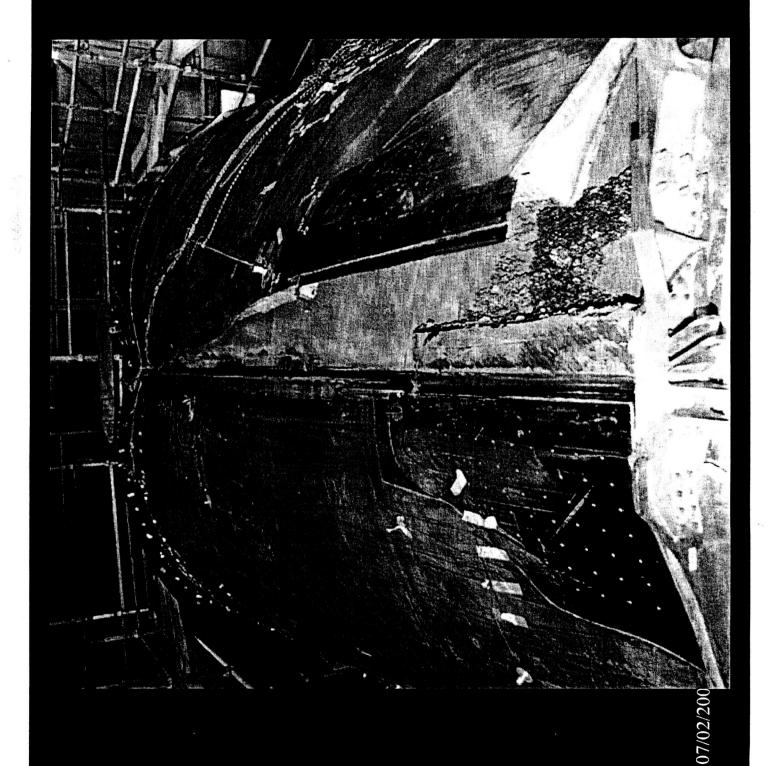
- September 21, 1999 test aborted due to hydrogen
- •100% cryogen fill (LH2)
- 20 psig internal pressure
- November 3, 1999 test completed
- •100% cryogen fill (LH2) at 42 psig internal **Dressure**
- Load case 5 applied at 5 psig internal pressure
- Tank drained of cryogen

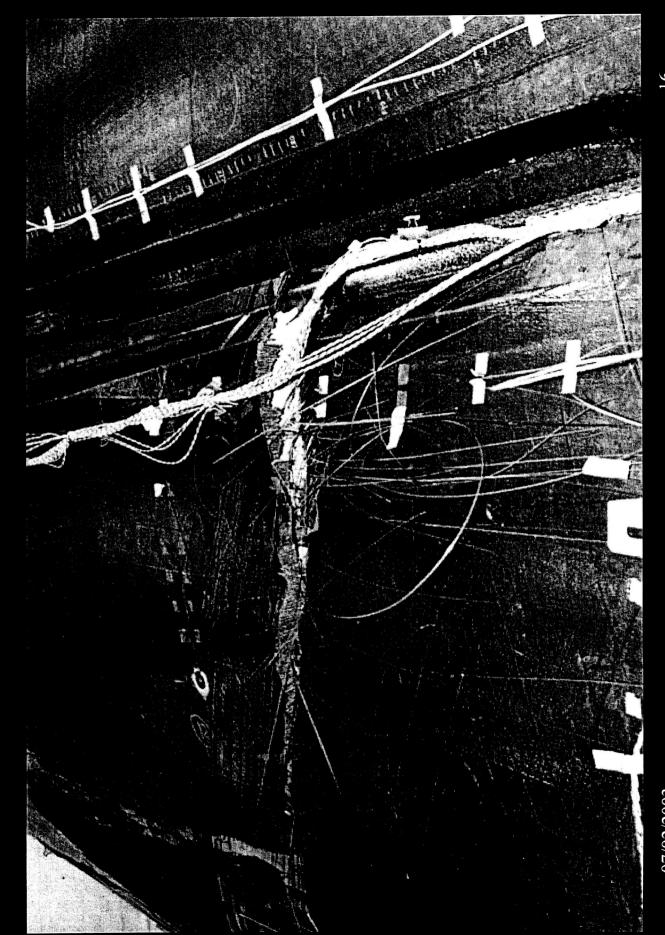
me

- Tank filled, 12:30 PM
- Tank pressurized to 42 psig, 2:00 PM
- Tank vented to 2 psig, 3:00 PM
- Loads applied, pressure increased to 5 psig, 4:40 PM
- Tank drained, 6:00 PM
- Lobe 1 failure, 6:24 PM

Camera 14: Lobe 1 and Lobe 4
Longeron

- Peel Failure
- Outer skin and core peeled away from inner skin
- Core Failure
- Core is 'mangled'
- Hydraulic fluid on test article
- Foreign Object Debris (FOD)
- Poor bondlines
- Pressure in core above ambient 13 hours after failure





07/02/2003





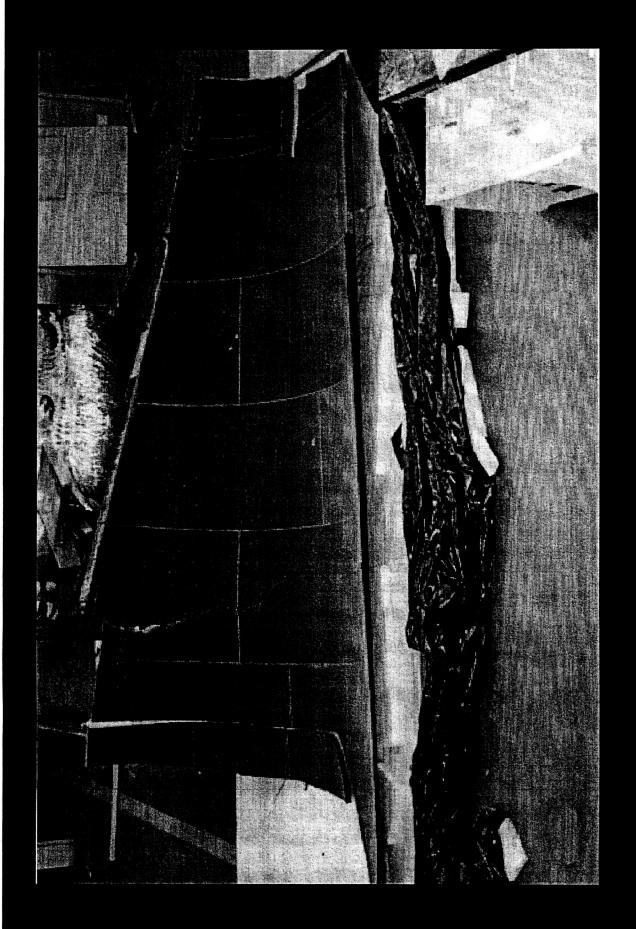


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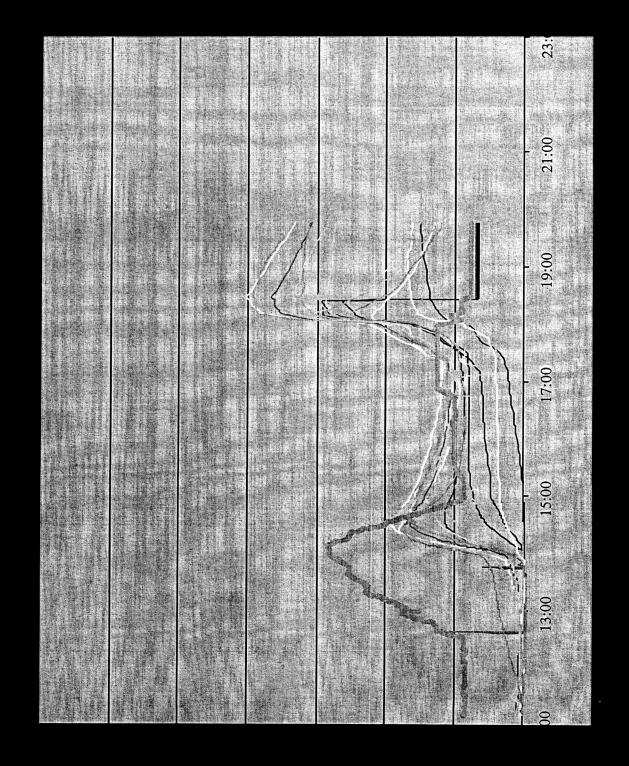


Subsequent Findings

Core pressures behaved unexpectedly

Microcracking

Core Pressure vs Time



Conclusions

The inner skin microcracked and hydrogen infiltrated

The cracks grew larger under pressure

When pressure was removed cracks closed slightly

 When tank was drained and warmed, cracks closed and blocked leak path

 FOD and debond areas provided an opportunity for a leak path

There is still hydrogen in the other 3 lobes today